

A LOW-COST POSITION AND VELOCITY ESTIMATION SYSTEM FOR MULTIROTOR AERIAL VEHICLES USING A STATIC KINECT SENSOR

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Abstract— Indoor estimation of position and velocity of a multirotor aerial vehicle (MAV) is frequently required for experimental evaluation of control laws and on-board navigation systems. The present work proposes a low-cost indoor position and velocity estimation system using a static Kinect sensor fixed on the ceiling and pointing downwards. The Kinect sensor consists of an RGB camera, an infrared projector and a CMOS camera. Using the RGB camera, it is possible to determine the horizontal position of the vehicle. On the other hand, the infrared projector together with the CMOS camera provide depth images which allows altitude estimation. The Kinect measurements are acquired in a desktop computer and processed in MATLAB. The data processing includes 3D position determination from a camera model and 3D position/velocity estimation using a Kalman filter. The proposed system is demonstrated and evaluated experimentally by means of the feedback control of an MAV.

Keywords— State estimation, multirotor aerial vehicles, Kinect sensor.

Resumo— Um sistema de estima  o de posi  o e velocidade de multic pteros em ambiente interior   frequentemente requerido para fins de avalia  o de leis de controle e algoritmos de navega  o. O presente trabalho prop e um tal sistema de baixo custo usando um sensor Kinect fixado ao teto e apontado verticalmente para baixo. O sensor Kinect   composto de uma c mera RGB, um emissor de sinal infravermelho e uma c mera CMOS. Mediante a c mera RGB,   poss vel determinar a posi  o horizontal do ve culo. A c mera CMOS, por sua vez, prov  uma imagem de profundidade baseada no tempo de viagem do sinal infravermelho emitido. Com essa  ltima imagem,   poss vel determinar a altitude de ve culo. As medidas do Kinect s o adquiridas num computador e processadas em MATLAB. O processamento de dados inclui a determina  o de posi  o 3D a partir do modelo do sensor, assim como estima  o de posi  o/velocidade 3D usando um filtro de Kalman. O sistema proposto   demonstrado e avaliado experimentalmente mediante do controle realimentado de um MAV.

Palavras-chave— Estima  o de estados, multic pteros, sensor Kinect.

1 Introduction

In the last few years, the world has seen an increasing interest for unmanned aerial vehicles (UAV) in both industry and academia. In particular, small multirotor aerial vehicles (MAV) have become very popular due to their simplicity, low cost, and capability to fly in confined areas. These vehicles are used in several applications, such as precision agriculture (Zhang and Kovacs, 2012), disaster monitoring (Alvissalim et al., 2012), search and rescue (Naidoo et al., 2011), surveillance (Grover, 2015), powerline inspection (Luque-Vega et al., 2014), aerial photography (Widodo and Sunardi, 2014), and building inspection (Emelianov et al., 2014; Bulgakov et al., 2014).

Many papers on control and navigation of MAVs have appeared in the literature in the last five years (Prado and dos Santos, 2014; Viana et al., 2015a; Viana et al., 2015b; dos Santos et al., 2013; Crowther et al., 2011; Mahony et al., 2012; Yan et al., 2013). In such works, experimental evaluation is always required to assess the performance of the proposed methods. In particular, indoor flight tests are more common for allowing an easier control of the flight environment. Some consolidated research groups have used ex-

pensive commercial tracking systems based on infrared cameras (Campbell et al., 2012; Ducard and D’Andrea, 2009; Zhou and Schwager, 2014).

The recent cost reduction of the so-called time-of-flight cameras has encouraged many research applications. In particular, the Microsoft Kinect sensor appears in many of those applications. Examples of Kinect applications are found in estimation of robot’s 3D position (Siradjuddin et al., 2012; El-laithy et al., 2012; Ganganath and Leung, 2012), in 3D object tracking (Chow et al., 2012), in medicine (Clark et al., 2012), and in education (Hsu, 2011), just to cite a few.

The present paper proposes a low-cost position and velocity estimation system for indoor evaluation of MAV guidance and navigation methods. The system consists of a Kinect sensor fixed on the ceiling and pointing downwards, and a computer for data processing. The MAV under test carries a circular red mark for providing contrast with respect to the floor (the background). The Kinect images are processed in MATLAB using color and circle segmentation for 3D position determination, as well as a Kalman filter for position smoothing and 3D velocity estimation. The proposed system is evaluated experimentally through static and flight tests with an MAV.

The remaining text is organized in the follow-

ing manner. Section 2 details the main components of the proposed estimation system. Section 3 details the implementation of the system. Section 4 evaluates the performance of the proposed system experimentally. Finally, Section 5 concludes the paper.

2 The System Components

The proposed low-cost 3D position/velocity estimation system consists of three main hardware components: a Kinect sensor fixed on the ceiling and pointing downwards; a ground computer for image processing and state estimation; and the object to be tracked, which, in particular, is an MAV. Figure 1 illustrates the overall system.

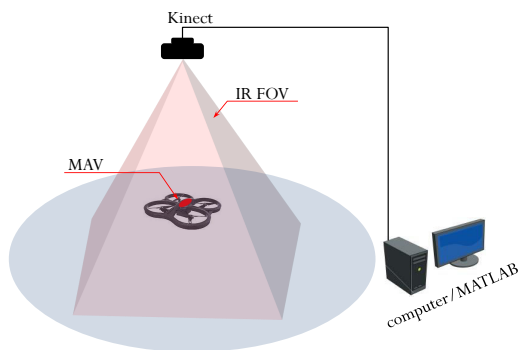


Figure 1: The overall position/velocity estimation system.

The following subsections describe the main components of the system.

2.1 Kinect Sensor

The Kinect sensor is a low-cost time-of-flight camera offered by Microsoft for the Xbox's video game consoles and for Windows operating system. It combines an RGB sensor with a depth sensor. The latter consists of an infrared (IR) projector and a monochrome CMOS sensor (Nakamura, 2011). In our estimation system, it provides the raw data sufficient for determining the 3D position of a red mark placed on the top of an MAV.

The Windows version of the Kinect sensor is adopted in the present work. It is installed on the ceiling at a height h from the floor and pointing downwards vertically. Its field of view (FOV) is of 53° about X_C and 47° about Y_C .

2.2 Multirotor Aerial Vehicle

The multirotor aerial vehicle (MAV) chosen for evaluating the proposed estimation system is a Parrot AR.Drone 2 quadrotor vehicle. It is a very popular low-cost MAV suitable for indoor operation. Originally, it was developed to be manually controlled by a smartphone applicative, but it can

also receive attitude and throttle commands via an Wi-Fi connection from an external computer. For more details about the navigation and control technology used in the AR.Drone, we suggest to read the reference (Bristeau et al., 2011).

In the present work, a circular red mark is attached on the top of the AR.Drone, as depicted in Figure 2. In this case, there is a sufficient contrast between the vehicle and the floor.



Figure 2: The AR.Drone with a circular red mark.

2.3 Ground Computer

The ground computer receives the Kinect data via an USB interface. The RGB and depth images are acquired and processed in MATLAB, using commands available in its Image Processing Toolbox. In short, from the depth image, we immediately obtain the vehicle's vertical coordinate z (altitude). On the other hand, the RGB images are segmented for separating the circular red mark from the background. The horizontal coordinates (x and y) of the mark's centroid are then computed in pixels. Using a pinhole camera model, this centroid is converted to millimeters. The x , y , and z are used as measurement in a Kalman filter (KF) that estimates 3D position and velocity. Finally, the ground computer delivers the KF position and velocity via UDP protocol to an external computer responsible for the automatic position control.

3 Position and Velocity Estimation

The implementation of the position and velocity estimation system is divided into three parts: image processing, position determination, and Kalman filtering.

3.1 Image Processing

The image processing is conducted in three steps in the following sequence:

1. Conversion of the color representation from the RGB model to the HSV model.

2. Image segmentation using thresholding in the HSV space. The output of this operation is a binary image where the original red pixels are converted into white and the other pixels are converted to black.
3. Circle recognition using the Hough transform. The output of this operation is the pixel coordinates of the black circle of the binary image generated in 2.

The above image processing methods are standard. For the readers who are interested in the details about them, we suggest reading the reference (Gonzalez et al., 2009).

3.2 Position Determination

Figure 3 illustrates an MAV and four Cartesian coordinate systems (CCS). The body CCS $S_B \triangleq \{X_B, Y_B, Z_B\}$ is attached to the vehicle's structure, with its origin at vehicle's center of mass (CM) and the Z_B axis perpendicular to the rotor plane and pointing upwards. The ground CCS $S_G \triangleq \{X_G, Y_G, Z_G\}$ is fixed on the ground with the origin at a known point O and the Z_G axis pointing upwards, aligned with the local vertical. The camera CCS $S_C \triangleq \{X_C, Y_C, Z_C\}$ is static with respect to S_G ; its origin is placed at the camera optical center C and its Z_C axis points downwards, parallel to the camera optical axis. Finally, the image CCS $S_I \triangleq \{X_I, Y_I\}$ is parallel to the X - Y plane of S_C and is centered at the camera principal point P .

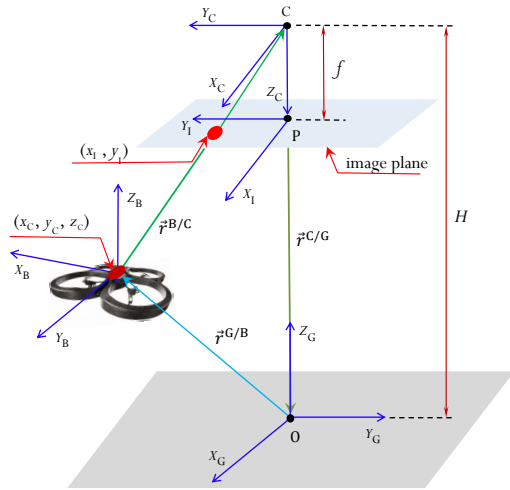


Figure 3: Cartesian coordinate systems and a pin-hole camera model.

Let the geometric vectors $\vec{r}^{B/C}$ and $\vec{r}^{B/G}$ describe the CM position w.r.t. S_C and S_G , respectively. On the other hand, let the geometric vector $\vec{r}^{C/G}$ describe the position of C w.r.t. S_G . From the figure, one can write

$$\vec{r}^{B/G} = \vec{r}^{C/G} + \vec{r}^{B/C}. \quad (1)$$

Denote the algebraic vectors representing $\vec{r}^{B/G}$ and $\vec{r}^{C/G}$ in S_G by $\mathbf{r}_G^{B/G}$ and $\mathbf{r}_G^{C/G}$, respectively. On the other hand, denote the algebraic vector representing $\vec{r}^{B/C}$ in S_C by $\mathbf{r}_C^{B/C}$. Moreover, denote the attitude matrix representing the attitude of S_C w.r.t. S_G by $\mathbf{D}^{C/G}$. Using the above notation, one can rewrite equation (1) as

$$\mathbf{r}_G^{B/G} = (\mathbf{D}^{C/G})^T \mathbf{r}_C^{B/C} + \mathbf{r}_G^{C/G}, \quad (2)$$

where $\mathbf{r}_G^{C/G} \triangleq [0 \ 0 \ h]^T$ is supposed to be known,

$$\mathbf{D}^{C/G} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \pi & -\sin \pi \\ 0 & \sin \pi & \cos \pi \end{bmatrix}, \quad (3)$$

which corresponds to a rotation of π rad about X_C , and $\mathbf{r}_C^{B/C} \triangleq [x_C \ y_C \ z_C]^T$ has to be computed from the output of the image processing.

Suppose that, from the image processing, it was found that the centroid of the red mark is located at (x_I, y_I) in S_I coordinates. On the other hand, consider that z_C is directly provided by the depth camera. Therefore, using a pinhole camera model (Hartley and Zisserman, 2003), the other two components of $\mathbf{r}_C^{B/C}$ can be computed by

$$x_C = \frac{z_C}{f} x_I, \quad (4)$$

$$y_C = \frac{z_C}{f} y_I, \quad (5)$$

where f is the camera focal length, which can be estimated using standard calibration packages.

3.3 Kalman Filtering

In this section, we design a discrete-time Kalman filter for estimating the MAV position $\mathbf{r}_G^{B/G} \triangleq [r_x \ r_y \ r_z]^T$ and velocity $\dot{\mathbf{r}}_G^{B/G} \triangleq [v_x \ v_y \ v_z]^T$. To begin with, define the state vector $\mathbf{x} \triangleq [r_x \ r_y \ r_z \ v_x \ v_y \ v_z]^T$. Now, we describe the position dynamics as the integral of a sum of velocity and velocity noise. On the other hand, we model the velocity dynamics as the integral of acceleration noise. Therefore, the continuous-time state-space model for $\mathbf{x}$ is given by

$$\dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{w}(t) \quad (6)$$

where $\mathbf{w} \in \mathbb{R}^6$ is the continuous-time state noise vector and

$$\mathbf{A} \triangleq \begin{bmatrix} \mathbf{0}_{3 \times 3} & \mathbf{I}_3 \\ \mathbf{0}_{3 \times 3} & \mathbf{0}_{3 \times 3} \end{bmatrix}. \quad (7)$$

The discrete-time state equation equivalent to (7) is immediately obtained as

$$\mathbf{x}_{k+1} = \Phi \mathbf{x}_k + \mathbf{w}_k, \quad (8)$$

where $\Phi \in \mathbb{R}^{6 \times 6}$ is the state transition matrix given by

$$\Phi = \begin{bmatrix} \mathbf{I}_3 & T_s \mathbf{I}_3 \\ \mathbf{0}_{3 \times 3} & \mathbf{I}_3 \end{bmatrix}, \quad (9)$$

with T_s representing the sampling time and $\mathbf{w}_k \in \mathbb{R}^6$ is the discrete-time state noise vector, which is assumed to be a zero-mean white Gaussian sequence with covariance $\mathbf{Q}_k$.

Assume that the initial state is a Gaussian vector with known mean $\bar{\mathbf{x}}$ and covariance $\bar{\mathbf{P}}$.

Now, define the filter measurement vector $\mathbf{y}$ to be the CM position as computed by equation (2), *i.e.*, $\mathbf{y} \triangleq \mathbf{r}_G^{\text{B/G}}$. Therefore, from the state definition, one can establish the following discrete-time measurement equation:

$$\mathbf{y}_{k+1} = \mathbf{C}\mathbf{x}_{k+1} + \mathbf{v}_{k+1}, \quad (10)$$

where $\mathbf{C} \in \mathbb{R}^{3 \times 6}$ is the measurement matrix given by

$$\mathbf{C} = \begin{bmatrix} \mathbf{I}_3 & \mathbf{0}_{3 \times 3} \end{bmatrix} \quad (11)$$

and $\mathbf{v}_{k+1}$ is the measurement noise, which is assumed to be a zero-mean white Gaussian sequence with known covariance $\mathbf{R}_{k+1}$.

Finally, using the state-space model given by (8) and (10), the discrete-time formulation of the Kalman filter algorithm (Bar-Shalom et al., 2004) can be immediately applied to optimally estimate the state vector $\mathbf{x}_k$ over time.

4 Results

The low-cost position and velocity estimation system proposed in this paper is now evaluated experimentally. For this end, two experiments are conducted: a static test and a flight test of feedback position control of an AR.Drone 2 quadrotor vehicle. The KF covariance matrices $\mathbf{P}_0$, $\mathbf{Q}_k$, and $\mathbf{R}_k$ are tuned with values $\mathbf{P}_0 = \text{diag} \{1.0 \times 10^{-2} \mathbf{I}_3, 1.0 \times 10^{-2} \mathbf{I}_3\}$, $\mathbf{Q}_k = \text{diag} \{1.0 \times 10^{-8} \mathbf{I}_3, 1.0 \times 10^{-7} \mathbf{I}_3\}$, and $\mathbf{R}_k = 1.0 \times 10^{-5} \mathbf{I}_3$. Using least-squares estimation, the camera focal length is calibrated in $f_x = 570$ and $f_y = 533$. The Kinect is installed at $H = 3.3$ m above the floor. The sampling time is set to $T_s = 0.08$ s.

4.1 Static Evaluation

The first test to evaluate the estimation system present in Section 2–3 is a static one. The target, which is a circular red mark with a diameter of 10 cm, is placed with its center at a given position $\mathbf{p}^i$ with zero velocity. The position $\mathbf{p}^i$ is then measured by means of a measuring tape and estimated with the proposed system. Denote the position measure by $\check{\mathbf{p}}^i$ and the corresponding estimate by $\hat{\mathbf{p}}^i$.

Table 1: Statistical test with 10 samples.

	$\mathbf{w}_1$	$\mathbf{w}_2$	$\mathbf{w}_3$	$\mathbf{w}_4$	$\mathbf{w}_5$
e_x	0.004	0.001	0.048	0.002	0.016
e_y	0.045	0.050	0.033	0.019	0.008
e_z	0.034	0.012	0.058	0.056	0.012
μ_x	0.004	0.501	-0.451	-0.497	0.484
μ_y	0.045	-0.800	-0.783	0.730	0.758
μ_z	0.734	0.687	0.758	0.756	0.712
σ_x	0.001	0.002	0.001	0.002	0.005
σ_y	0.001	0.002	0.005	0.002	0.006
σ_z	0.007	0.008	0.005	0.008	0.009

Define the following metric to evaluate the system performance regarding its capability to estimate position:

$$e_k^i = \|\check{\mathbf{p}}_k^i - \hat{\mathbf{p}}_k^i\|. \quad (12)$$

The above test is repeated to obtain 10 samples of e_k^i . The sample mean μ^i and standard deviation σ^i are registered in Table 1. The test is realized at five different positions, corresponding to $i = 1, 2, \dots, 5$.

where it is maintained static validation of the positions estimated by the proposed system were made during tests with metric tapes. For this evaluation, it was taken 10 measurements for five desired variables of position (waypoints) in a 3D indoor environment. The test waypoints are defined as: $\mathbf{w}_1 = [0 \ 0 \ 0.7]^T$ m, $\mathbf{w}_2 = [0.5 \ -0.75 \ 0.7]^T$ m, $\mathbf{w}_3 = [-0.5 \ -0.75 \ 0.7]^T$ m, $\mathbf{w}_4 = [-0.5 \ 0.75 \ 0.7]^T$ m, $\mathbf{w}_5 = [0.5 \ 0.75 \ 0.7]^T$ m. The precision of this test are showed in Table 1 which includes summary statistics such as the mean μ , standard deviation σ , and the error between the desired values and the mean position of each dimension provided by the Kalman Filter output.

As the reader can see in Table 1, estimated position errors stay below 2 cm in most samples, and is around 5 cm in the worst case, that is acceptable for our application.

4.2 Closed-Loop Validation

In this subsection, experimental results validate the proposed estimation system by using a position controller to guide the AR.Drone quadrotor in the environment. The waypoints chosen for the flight test are trajectories composed by a set of straight lines from the initial position $\mathbf{r}_i = [0 \ 0 \ 0.5]^T$ passing by the waypoints $\mathbf{w}_1 = [0.4 \ 0 \ 0.5]^T$, $\mathbf{w}_2 = [0.4 \ -0.5 \ 0.5]^T$, $\mathbf{w}_3 = [-0.4 \ -0.5 \ 0.5]^T$, $\mathbf{w}_4 = [-0.4 \ 0.5 \ 0.5]^T$, $\mathbf{w}_5 = [0.4 \ 0.5 \ 0.5]^T$. The position controller is composed by the altitude and horizontal controllers. The altitude control is a Proportional (P) controller and the horizontal control is a Proportional-Derivative (PD) controller. The constant velocity chosen in the trajectory generation was $v = 0.5$ m/s².

Analyzing the flight data obtained through the proposed system, Figure 4 shows the innovation e_k during the flight test. As it can be seen in Figure 4 the variable e_k tends to converge to zero, except for the waypoint switching transients.

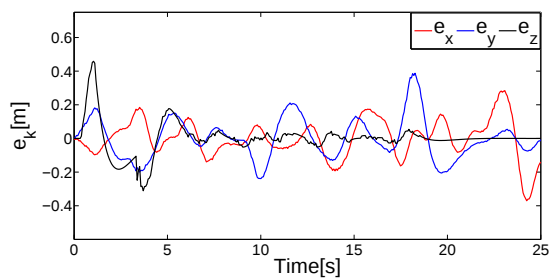


Figure 4: Innovation in a flight test.

For evaluation of the position estimation were generated separate graphics for each position state comparing with their desired values, see Figure 5. Note that the estimated positions are well close to the waypoints references, where the position error oscillate by $\pm 0.02m$ around their desired values, that is a good precision.

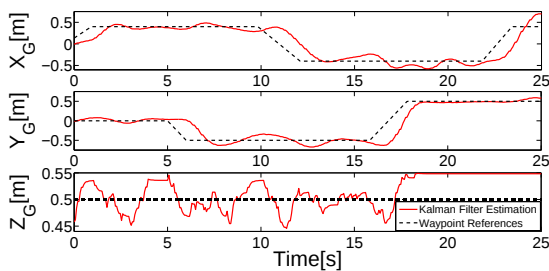


Figure 5: Position state estimation.

5 Conclusions

This paper demonstrated a low-cost alternative for the estimation of velocity and position of a MAV in indoor flight. To develop this solution it was used three main components: a low-cost Kinect sensor; a MAV; and a ground computer equipped with MATLAB. First, from the sensor data provided, the computer vision methods were used to identify and tracking a round red color mark attached over the AR.Drone 2 quadrotor. To find the round color mark and determine its centers the Hough transform (HT) was used. After the centroid of the color mark be well detected, to determine its position coordinates in S_G from the image coordinates S_I , we used geometric transformation as detailed in section 3. In sequence, to estimate the velocities, we aided a Kalman filter as output of the system.

Finally, the proposed system was evaluated through a static experiment and by using a closed-loop validation. In both, the experimental results

showed a good performance of the estimation system, where the errors in the position estimation oscillated within a range allowed for our application, which is intended to validate guidance and navigation methods in the future.

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